NTSB ID: FTW02FA262 Aircraft Registration Number: N6227J

Occurrence Date: 09/23/2002 Most Critical Injury: Fatal

Occurrence Type: Accident Investigated By: NTSB

Location/Time

Nearest City/Place	State	Zip Code	Local Time	Time Zone	
Chromo	CO 81128 2		2015	MDT	
Airport Proximity: Off Airport/Airstrip	Distance Fror	n Landing Facility:		Direction Fro	m Airport:

Aircraft Information Summary

7 morale morniation Garmay		
Aircraft Manufacturer	Model/Series	Type of Aircraft
Piper	PA-28R-200	Airplane

Sightseeing Flight: No Air Medical Transport Flight: No

#### Narrative

 $Brief\ narrative\ statement\ of\ facts,\ conditions\ and\ circumstances\ pertinent\ to\ the\ accident/incident:$ 

HISTORY OF FLIGHT

On September 23, 2002, approximately 2015 mountain daylight time (all times are mountain daylight time unless otherwise noted), a Piper PA-28R-200 single-engine airplane, N6227J, was destroyed when it impacted mountainous terrain during cruise flight near Chromo, Colorado. The flight instructor and instrument-rated private pilot, who was receiving instruction, sustained fatal injuries. The airplane was owned and operated by Melcarl Aviation Company of Parker, Colorado. Dark night visual meteorological conditions prevailed, and a flight plan was not filed for the 14 Code of Federal Regulations Part 91 instructional flight. The cross-country flight departed the Fullerton Municipal Airport, Fullerton, California, approximately 1345 pacific daylight time (PDT), with planned fuel stops at Prescott, Arizona, and Farmington, New Mexico, and was destined for the Centennial Airport (APA), Denver, Colorado.

In a telephone interview conducted by the NTSB investigator-in-charge (IIC), the owner of Gerdes Aviation Services (an aviation brokerage company, in Fullerton, California) stated on the afternoon of September 22, 2002, he picked up the flight instructor and private pilot from the Ontario International Airport, Ontario, California. The two pilots, who were the president and vice president of Melcarl Aviation Company, had flown on a commercial flight from Denver to Ontario. The pilots checked into their hotel, and then the three of them went to the airport to inspect the Piper airplane. The flight instructor, who was also an airframe and powerplant mechanic, inspected "everything." The two pilots and another individual then test flew the airplane, and after the test flight, the two pilots were "happy with the airplane." One discrepancy was noted during the test flight; the rear vent was not functional. It was determined the flapper valve nut was loose; the nut was tightened, and the discrepancy was eliminated. The brokerage owner then drove the pilots back to their hotel.

The following day, also the day of the accident, at 0700 PDT, the pilots called their financial institution regarding the financing for the airplane. At 0900 PDT, the brokerage company owner picked the two pilots up from the hotel, and all three went to the brokerage owner's financial institution to complete a portion of the purchase transaction. At 1100 PDT, the three individuals arrived at the Fullerton Municipal Airport. The flight instructor and private pilot loaded the airplane with some personal items, fueled the airplane, and then departed the airport approximately 1345 PDT. According to the brokerage owner, the pilots wanted to leave Fullerton by 0900 PDT; however, there were some problems encountered with the financing that had to be resolved that morning.

The brokerage owner further stated the flight instructor was seated in the right seat, and the private pilot was seated in the left seat. The private pilot was going to fly the entire trip as a flight checkout in the accident airplane.

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According to a fuel receipt, at 1509 mountain standard time, the airplane was fueled with 25.30 gallons of 100LL aviation fuel at the Prescott Municipal Airport, Prescott, Arizona.

According to communications provided by the Farmington Control Tower, the airplane landed at the Four Corners Regional Airport (FMN), Farmington, New Mexico, at 1903. At 1910, the airplane was fueled with 20.2 gallons of 100LL aviation fuel at the SevenBar Four Corners fixed base operator (FBO). At 1916, one of the two pilots contacted the Albuquerque Automated Flight Service Station (AFSS), via telephone, to obtain a weather update for the flight from FMN to APA. At 1933:42, the pilot was cleared for takeoff for an eastbound departure. At 1937:09, the pilot requested a frequency for flight following, and the controller instructed the pilot to contact Denver Center on frequency 118.57.

According to information provided by the Denver Air Route Traffic Control Center (ARTCC), at 1943:17, the pilot contacted Denver ARTCC. At 1943:39, the pilot reported that his position was approximately 4 miles west of the Farmington VOR (very high frequency omni directional range station), on victor airway 210, at an altitude 9,500 feet, and the pilot requested visual flight rules (VFR) flight following. At 1944:40, the controller requested the airplane type, and the pilot acknowledged the request. At 1945:00, the controller reported the current Farmington and Alamosa, Colorado, altimeter settings. At 1945:09, the pilot acknowledged the controller's transmission, which was the last radio communication from the airplane. At 2014:31, Denver ARTCC lost radar contact with the airplane and attempts to contact the airplane were unsuccessful. The last recorded radar return of the accident airplane was at 10,700 feet msl.

Approximately 2141, an Alert Notice (ALNOT) was issued by the Denver ARTCC for the accident airplane. At 2214, the Air Force Rescue Coordination Center received an emergency locator transmission (ELT), and at 2227, they contacted the Civil Air Patrol (CAP), of Durango, Colorado. A search was initiated by the CAP, and at 0730, on September 24, 2002, the airplane wreckage was located at 37 degrees 6 minutes 5 seconds North latitude and 106 degrees 44 minutes 23 seconds West longitude. The accident site was approximately 85 miles northeast of FMN.

# PERSONNEL INFORMATION

The flight instructor held a commercial pilot certificate with airplane multiengine land and sea, airplane single-engine land and sea, glider, and instrument airplane ratings. The flight instructor held a certified flight instructor certificate for airplane single-engine land, airplane multiengine land, glider, and instrument airplane. In addition, the flight instructor also held an airframe and powerplant certificate, an inspection authorization certificate, and a flight engineer certificate. The instructor was issued a second class medical certificate on March 15, 2002, with the limitation, "Holder shall wear lenses that correct for distant vision while exercising the privileges of his/her airman certificate." The flight instructor's logbook was not located; however, according to his most recent medical application, the instructor reported having accumulated a total flight time of 9,000 hours, of which, 500 hours were flown in the preceding six months.

In a telephone interview conducted by the NTSB IIC, a former student of the flight instructor reported it was "uncharacteristic for [flight instructor] to fly low over mountains." He added the flight instructor was extremely cautious, and had flown several night cross-country flights over the mountains.

The private pilot was issued his private pilot certificate on May 24, 2000, and was issued his instrument airplane rating on July 3, 2002. The private pilot was issued a first class medical certificate on March 25, 2002, with no limitations or restrictions. A review of the private pilot's logbook revealed that he had accumulated approximately 272 total flight hours. In the preceding 90 days, the pilot had logged 27 hours total flight time, and in the preceding 30 days, he had logged 15 hours of flight time. The private pilot had received most of his flight

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Narrative (Continued)

instruction with the flight instructor.

#### AIRCRAFT INFORMATION

The 1976-model Piper PA-28R-200 airplane, serial number 28R-7635323, was a low wing, retractable landing gear, semi-monocoque design airplane. The airplane was powered by a four cylinder, air-cooled, horizontally opposed, normally aspirated Lycoming IO-360-C(1)C (serial number L-15663-51A) engine, rated at 200 horsepower. The airplane was configured to carry a maximum of four occupants.

The airplane was issued a standard airworthiness certificate and was certificated for normal category operations. The airplane was registered to Melcarl Aviation Company on the day of the accident. A review of the maintenance records revealed the airframe underwent its most recent annual inspection on September 16, 2002, at a tachometer time of 833.30 hours and a total time in service of 4,851.80 hours. In addition, the most recent 100-hour inspection on the engine was completed on September 16, 2002, and the engine had accumulated 1,166.3 hours since its last major overhaul. The tachometer indication at the accident site was 841.2 hours.

An "Aircraft Purchase Agreement", which was signed and dated on September 13, 2002, by the brokerage company owner and the private pilot, was located in the airplane. According to the agreement, within 10 days of the agreement, several discrepancies were to be completed and repaired. The agreement stated the "buyer" would perform an acceptance flight before the purchase transaction was completed.

The airplane was equipped for instrument flight rules (IFR) operations, and an Arnav Star 5000 global positioning system (GPS) was installed on the airplane. The airplane was not equipped with an oxygen system or supplemental oxygen.

## METEOROLOGICAL INFORMATION

According to witnesses, who were located near the accident site at the time of the accident, the weather was reported as clear sky, and the wind was calm.

At 1953, the Durango, Colorado, automated surface observing system (ASOS), located approximately 45 nautical miles west of the accident site, reported the wind from 210 degrees at 6 knots, visibility 10 statute miles, sky clear, temperature 64 degrees Fahrenheit, dew point 18 degrees Fahrenheit, and an altimeter setting of 30.20 inches of mercury.

At 2014, the Alamosa ASOS, located approximately 46 nautical miles northeast of the accident site, reported the wind from 290 degrees at 11 knots and gusting to 35 knots, visibility 2 1/2 statute miles in decreasing rain, few clouds at 5,000 feet, sky broken at 6,500 feet and overcast at 10,000 feet, temperature 64 degrees Fahrenheit, dew point 32 degrees Fahrenheit, and an altimeter setting of 30.29 inches of mercury.

The U.S. Naval Observatory reported that the moonrise on the night of the accident for Pagosa Springs, Colorado, located approximately 15 miles northwest of the accident site, was at 2024.

# WRECKAGE AND IMPACT INFORMATION

The airplane impacted heavily wooded mountainous terrain at an elevation of 11,100 feet msl. The wreckage energy path measured approximately 75 feet in length on a measured magnetic heading of 060 degrees. The airplane came to rest upright on an estimated 60 degree slope, and the fuselage displayed forward-to-aft accordion type crush damage. Several trees (varying in diameter from 6 to 12 inches), located at the start of the wreckage energy path to the wreckage, displayed fresh breaks and fractures. The tree breaks and fractures were at a level height and attitude. There

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were no ground scars noted expect for the scars beneath the main wreckage.

The outboard 4 feet sections of the left and right wings were found separated and distributed along the wreckage path. The outboard 1 foot of the left horizontal stabilator was separated, and the horizontal stabilator remained attached to the empennage. The vertical stabilizer and rudder remained intact and attached to the empennage.

The cockpit and cabin area were destroyed. The left and right forward seats were removed by the search and rescue personnel. Both control yokes were separated from their respective control tubes, and the left control horns were separated on both control yokes. The engine remained attached to the engine mounts and airframe. The three blade propeller assembly was found separated from the engine crankshaft, and two blades were separated from the propeller hub.

The wreckage was recovered to Air Transport, Phoenix, Arizona, for further examination.

#### PATHOLOGICAL INFORMATION

Both autopsies were performed by the Archuleta County Coroners Office, Montrose, Colorado, on September 27, 2002. Specimens for the toxicological tests were taken from both pilots.

The Federal Aviation Administrations (FAA) Civil Aeromedical Institute's (CAMI) Forensic and Accident Research Center examined the specimens taken by the medical examiner. According to CAMI, testing for carbon monoxide and cyanide were not performed. The private pilot's toxicology showed no indication of performance-impairing drugs at the time of the accident.

According to CAMI, the flight instructor's toxicological tests indicated 0.226 (ug/ml, ug/g) of Diphenhydramine (commonly known by the trade name Benadryl) was detected in the blood and an undetermined amount was detected in the liver. Diphenhydramine is an over-the-counter antihistamine with sedative effects, often used to treat allergy symptoms.

#### TEST AND RESEARCH

On January 29 and 30, 2003, at the facilities of Air Transport and under the supervision of the NTSB IIC, representatives from The New Piper Aircraft and Lycoming examined the wreckage. The examination of the airframe revealed the left wing was separated at the airframe attach point. The leading edge displayed several semi-circular indentations consistent with tree strikes. The left fuel cell was destroyed and fragmented. The left aileron and flap were separated and bent upward. The left aileron cable exhibited broomstrawing signatures which were consistent with overload. The right wing leading edge was separated from the main spar, and the main spar was separated outboard of the landing gear attach point. The right wing displayed crushing from the leading edge to the trailing edge of the wing. The right fuel cell was destroyed and fragmented. The right aileron remained attached to its respective hinge points, and the right flap was separated and displayed damage consistent with a tree strike. Control continuity to the right aileron was established.

The empennage section was separated from the fuselage during the recovery process. The horizontal stabilator and vertical stabilizer remained attached to the empennage. The horizontal stabilator link assembly was bent and separated at the control tube assembly. The fracture surfaces of the control tube assembly at the rod end displayed 45-degree sheer lips. The rudder and elevator trim tab control surfaces remained attached to their respective attach points. Control continuity of the rudder and elevator trim was established. The elevator trim drum was measured and determined to be in a slightly nose up position.

The cockpit and cabin floor was crushed aft toward the empennage. The flight and engine instruments were separated from their respective mounting structure and destroyed. The cockpit throttle, mixture, and propeller controls were found in the full forward, midrange, and midrange

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positions, respectively. The fuel selector was found in the RIGHT tank position. The "Comm 1" frequency indication was 118.57, which is a Denver ARTCC frequency, the "Nav 1" frequency indication was 115.30, which is a FMN VOR frequency. The transponder was destroyed. The flap selector pin was broken and the selector was found in the 20-degree detent position. The three-point altimeter instrument was destroyed; however, the face plate and pointers displayed an altitude of 8,600 feet and the Kollsman Window was set at 30.24 inches of mercury.

The left and right forward seats displayed forward to aft crushing, and all four seats were separated from their respective seat tracks. The left seatbelt was found latched and the shoulder harness remained attached to the lap belt. The right seatbelt was found unlatched and the shoulder harness remained attached to the lap belt. The door handle was found in the latched position and the door pin was in the engaged position.

The engine remained attached to the airframe by the engine mounts. The engine crankcase displayed several cracks and fractures forward of the #2 and #4 cylinders. Engine continuity was established from the crankshaft to the valve train and accessory gears when the crankshaft was rotated by hand. Thumb compression and suction was obtained on all four cylinders. The spark plugs were removed and according to the Champion Spark Check-A-Plug chart AV-27, the spark plug electrodes displayed coloration consistent with normal combustion. The magnetos and the ignition harnesses were destroyed and unable to be functionally tested. The vacuum pump was secured to its mounts, and the vacuum pump cover was removed. The vacuum pump rotor and vanes were found cracked in several locations. The engine exhaust and intake system was crushed and destroyed.

The propeller hub was separated from the engine crankshaft. Two of the blades were separated from the fractured hub and one blade remained in the propeller hub. Two of the blades displayed "S" bending and leading edge gouging, and one blade was bent toward low pitch.

### ADDITIONAL INFORMATION

The wreckage was released to the owner's representative January 30, 2003.

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AVIATION	rrence Type	: Accident	ccident									
Landing Facility/Approach In	formation											
Airport Name	Airport ID:	Airport Elev	ation ft. MSL		way Used	Runwa	y Lengti	h R	unway Width			
Runway Surface Type: Unknown												
Runway Surface Condition: Unknown	own											
Type Instrument Approach: Unknown	own											
VFR Approach/Landing: Unknown	١											
Aircraft Information												
Aircraft Manufacturer Piper				l/Series 28R-200					Number 7635323			
Airworthiness Certificate(s): Normal												
Landing Gear Type: Retractable	- Tricycle											
Homebuilt Aircraft? No	omebuilt Aircraft? No Number of Seats: 4						2650	LBS	Numbe	mber of Engines: 1		
Engine Type: En Reciprocating Ly				anufacturer: g			Model/Se IO-360-			Rated Power: 200 HP		
- Aircraft Inspection Information												
Type of Last Inspection			Date of La	st Inspection		Time Si	ne Since Last Inspection				Airframe Total Time	
Annual			09/2002					7.9 Ho	ours	4851.8 Hours		
- Emergency Locator Transmitter (	ELT) Information											
ELT Installed? Yes	ELT Operat	ted? Ye	s		ELT	Γ Aided i	n Locating Ac	cident S	ite? Ye	s		
Owner/Operator Information												
Registered Aircraft Owner			Street	t Address 12636 S. Oak Creek Ct.								
Melcarl Aviation Company			City	Parker	Parker C						Zip Code 80134	
			Street /	Address						100	100101	
Operator of Aircraft				Same a	s Reg	'd Aircr	aft Owner				T =	
Same as Reg'd Aircraft Owner	City						State	Zip Code				
Operator Does Business As:						0	perator Desig	nator Co	ode:			
- Type of U.S. Certificate(s) Held: I	None											
Air Carrier Operating Certificate(s)	:											
Operating Certificate: Operator Certificate:												
Regulation Flight Conducted Under: Part 91: General Aviation												
Type of Flight Operation Conducted	d: Positioning											
		FACTU	JAL REPO	ORT - AVIA	ΓΙΟΝ						Page 2	

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	AVIATI	Occurrence Type: Accident					1								
First Pilot	t Information								ļ						
Name					Τ	City				St	ate	Date of Birth	Age		
On File On										Or	n File	On File	58		
Sex: M	Seat Occupied				Certifica	ate Num	ber: On File	•							
Certificate(s): Commercial; Flight Engineer															
Airplane Rating(s): Multi-engine Land; Multi-engine Sea; Single-engine Land; Single-engine Sea															
Rotorcraft/Glider/LTA: Glider															
Instrument Rating(s): Airplane															
Instructor Rating(s): Airplane Multi-engine; Airplane Single-engine; Glider; Instrument Airplane															
Type Rating/Endorsement for Accident/Incident Aircraft? Yes  Current Biennial Flight Review?															
Medical Ce	rt.: Class 2	Medica	al Cert. Status	: Valid Me	dicalw/ wa	aivers/li	m.		Date of	e of Last Medical Exam: 03/2002  Rotorcraft Glider Lighter Than Air					
		<b> </b>													
- Flight Tim	Flight Time Matrix  All A/C  This Make and Model			Airplane Single Engine	Airplane Mult-Engine	Nigl	Night Ins		Instrument Simulated		Rotorcraft	Glider			
Total Time		9000													
Pilot In Cor	nmand(PIC)														
Instructor															
Last 90 Day	ys														
Last 30 Da	ys														
Last 24 Ho	urs					1									
Seatbelt Us	sed? Yes	Shou	lder Harness	Used? Yes			Toxico	logy Per	ormed? \	'es	s	econd Pilot?	Yes		
	n/Itinerary														
Departure F	ght Plan Filed: No	one				Т	State	Τ,	irport Ider	atifior	Dono	erturo Timo	Time Zone		
Farmingto							NM		MN	illiei	Departure Time 1934		TO		
Destination	ı .						State		irport Ider	ntifier					
Denver									.PA	itinoi					
Type of Cle	earance: VFR F	light Follow	ing												
Type of Air	space: Class	G													
Weather	Information														
Source of Briefing: Flight Service Station															
Method of	Method of Briefing: Telephone														
				FACTUAL	REPORT	- AVIA	ATION	1					Page 3		

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	ETYBOR		Occurrer	ice Type:	Accide	nt											
Weather	Weather Information																
WOF ID	Observation Time	Time Zone	WOF Eleva	tion	WOF D	WOF Distance From Accident Site Direction					rection From Accident Site						
FMN	1953	MDT	5506 F	t. MSL	85 NM 250 Deg.						eg. Mag.						
Sky/Lowes	st Cloud Condition: Clear				Ft. /	AGL	Condition o	of Ligh	nt: Dusk								
Lowest Ce	eiling: None		F	t. AGL	Visik	oility:	10	SM	Alti	meter:	30.04	"Hg					
Temperatu	ure: 22 °C [	Dew Point:	-7 °C	Wind	Direction	ા: Variable	<del>,</del>		Dei	nsity Altitude:		Ft.					
Wind Spee	∍d: Calm	Gusts:		Weatl	her Cond	tions at Ac	cident S	Site: Visual C	Cond	itions	250 Deg. Mag.  Dusk  eer: 30.04 "Hg  y Altitude: Ft.  ns						
Visibility (F	RVR): Ft.	Visibility	(RVV)	SM	Intensi	ty of Precip	pitation:										
Restriction	Restrictions to Visibility: None																
Type of Pro	Type of Precipitation: None																
Accident	Information																
Aircraft Da	mage: Destroyed		Aircraft Fi	re: None	<del></del>			Aircraft Exp	losio	n None							
Classificati	ion: U.S. Registered/U.	.S. Soil															
- Injury Su	mmary Matrix	Fatal	Serious Mir	ior	None	TOTAL											
First Pi	ilot	1				T	1										
Second	d Pilot	1					1										
Studen	nt Pilot						$\neg$										
Flight I	nstructor						7										
Check	Pilot																
Flight E	Engineer						$\neg$										
Cabin A	Attendants						$\neg$										
Other C	Crew						7										
Passen	ngers																
- TOTAL A	ABOARD -	2					2										
Other 0	Ground						7										
- GRANE	O TOTAL -	2					2										

National Transportation Safety Board

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# Administrative Information

Investigator-In-Charge (IIC)

Aaron M. Sauer

Additional Persons Participating in This Accident/Incident Investigation:

James E Mack Aviation Safety Inspector Federal Aviation Administration Denver, CO 80249

Charles Little Air Safety Investigator The New Piper Aircraft Chino Hills, CA 91709

John Butler Air Safety Investigator Lycoming Arlington, TX 76011